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#### **MECHANICS OF MATERIALS**

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Level YEAR 2

#### Statically Indeterminate Members:

When the reactive forces or the internal resisting forces over a cross section exceed the number of independent equations of equilibrium, the structure is called statically indeterminate. These cases require the use of additional relations that depend on the elastic deformations in the members.

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- If the equilibrium equations are sufficient to calculate all the forces (including support reactions) that act on a body, these forces are said to be *statically determinate*.
- □ In *statically determinate problems*, the number of unknown forces is always equal to the number of independent equilibrium equations.
- If the number of unknown forces exceeds the number of independent equilibrium equations, the problem is said to be *statically indeterminate*.
- A statically indeterminate problem always has geometric restrictions imposed on its deformation. The mathematical expressions of these restrictions known as the *compatibility equations*, provide us with the additional equations needed to solve the problem.
- Because the source of the compatibility equations is deformation, these equations contain as unknowns either strains or elongations. Use Hooke's law to express the deformation measures in terms of stresses or forces. The equations of equilibrium and compatibility can then be solved for the unknown forces (force-displacement equation).
- Procedure for Solving Statically Indeterminate Problem
- Draw the required free-body diagrams and derive the equations of equilibrium.
- Derive the compatibility equations. To visualize the restrictions on deformation, it is often helpful to draw a sketch that exaggerates the magnitudes of the deformations.
- Use Hooke's law to express the deformations (strains) in the compatibility equations in terms of forces (or stresses)
- Solve the equilibrium and compatibility equations for the unknown forces (force-displacement equation).



 $\frac{\delta_{st}}{0.6} = \frac{\delta_{br}}{1.6}$ 

#### Sample problem 1

Figure (a) shows a rigid bar that is supported by a pin at *A* and two rods, one made of steel and the other of bronze. Neglecting the weight of the bar, compute **the stress** in each rod caused by the 50-kN load, the following data :



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#### Solution

Equilibrium in Fig. (b),	$P_{\rm st}$		R <sub>br</sub>
contains four unknown forces.	A <sub>y</sub> 0.6 m	1.0 m	0.8 m
Since there are only three			
independent equilibrium	$A_{x}$		50 kN
equations, these forces are		(b) FBD	
statically indeterminate. The			
equilibrium equation that does	<u>4 0.6 m</u>	1.0 m	0.8 m
not involve the pin reactions at	$\delta_{st}$		$\delta_{br}$
A is		(c)	

$$\Sigma M_{\rm A} = 0 + 0.6P_{\rm st} + 1.6P_{\rm br} - 2.4(50 \times 10^3) = 0$$
 (a)

*Compatibility* The displacement of the bar, consisting of a rigidbody rotation about A, is shown greatly exaggerated in Fig. (c).

From similar triangles, we see that the elongations of the supporting rods must satisfy the compatibility condition

$$\frac{\delta_{st}}{0.6} = \frac{\delta_{br}}{1.6} \tag{b}$$

Hooke's law When we 0.6 m 1.0 m substitute  $\delta = PL/(EA)$  into Eq. 0.8 m (b), the compatibility equation becomes  $\frac{1}{0.6} \left(\frac{PL}{EA}\right)_{st} = \frac{1}{1.6} \left(\frac{PL}{EA}\right)_{br}$ 50 kN (b) FBD Using the given data, we obtain 0.6 m 1.0 m 0.8 m  $\frac{1}{0.6} \frac{P_{st}(1.0)}{(200)(600)} = \frac{1}{1.6} \frac{P_{br}(2)}{(83)(300)}$  $\delta_{br}$ δst (c)  $P_{\rm st} = 3.614 P_{\rm br}$  (c)

Note that the we did not convert the areas from  $mm^2$  to  $m^2$ , and we omitted the factor 10<sup>9</sup> from the moduli of elasticity. Since these conversion factors appear on both sides of the equation, they would cancel out.

Solving Eqs. (a) and (c), we obtain

$$P_{\rm st} = 115.08 \times 10^3 \,\rm N$$
  $P_{\rm br} = 31.84 \times 10^3 \,\rm N$ 

The stresses are

$$\sigma_{st} = \frac{P_{st}}{A_{st}} = \frac{115.08 \times 10^3}{600 \times 10^{-6}} = 191.8 \times 10^6 Pa = 191.8 MPa \quad Answer$$
  
$$\sigma_{st} = \frac{P_{br}}{A_{br}} = \frac{31.84 \times 10^3}{300 \times 10^{-6}} = 106.1 \times 10^6 Pa = 106.1 MPa \quad Answer$$

# Sample Problem 2Bearing<br/>plateThe concrete post in Fig. (a) is<br/>reinforced axially with four<br/>symmetrically placed steel bars, each<br/>of cross-sectional area 900 mm².<br/>Compute the stress in each material<br/>when the 1000-kN axial load is<br/>applied. The moduli of elasticity are<br/>200 GPa for steel and 14 GPa for<br/>.concreteBearing<br/>plateSteel -<br/>ConcreteSteel -<br/>Concrete



1000 kN

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$$\delta = \frac{\sigma L}{E} = \frac{PL}{EA} \qquad \qquad \frac{\sigma_{st}}{E_{co}} = \frac{\sigma_{co}}{E_{co}}$$
(c)

Equations (a) and (c) can now be solved for the stresses. From Eq. (c) we obtain

$$\sigma_{st} = \frac{E_{st}}{E_{co}} \sigma_{co} = \frac{200}{14} \sigma_{co} = 14.286 \sigma_{co} \qquad \text{(d)}$$

Substituting the cross-sectional areas  $A_{st} = 4 (900 \times 10^{-6}) = 3.6 \times 10^{-3} \text{ m}^2$   $A_{co} = 0.3^2 - 3.6 \times 10^{-3} = 86.4 \times 10^{-3} \text{ m}^2$   $4P_{st} + P_{co} - 1.0 \times 10^6 = 0$ and Eq. (d) into Eq. (a) yields  $Than \ \sigma_{st} = P_{st} / A_{st}$   $(14.286 \ \sigma_{co})(3.6 \times 10^{-3}) + \sigma_{co}(8.64 \times 10^{-3}) = 1.0 \times 10^6$ Solving for the stress in concrete, we get  $\sigma_{co} = 7.255 \times 10^6 \text{ Pa} = 7.255 \text{ MPa}$  AnswerFrom Eq. (d), the stress in steel is  $\sigma_{st} = 14.286 (7.255) = 103.6 \text{ MPa}$  Answer

#### Sample Problem 3

Let the allowable stresses in the post described in Problem 1 be  $\sigma_{st}$  =120 MPa and  $\sigma_{co}$  = 6 MPa. Compute the maximum safe . axial load *P* and may be applied

#### Solution

substituting the allowable stresses into the equilibrium equation—see Eq. (a) in Sample Problem 2.6. This approach is **incorrect** because it ignores the compatibility condition, the equal strains of , the two materials,  $\delta_{st} = \delta_{co}$ , From Eq. (d) in Sample problem 2.6  $\sigma_{st} = 14.286 \sigma_{co}$  (the concrete were broked rather than steel) Therefore, if the concrete were stressed to its limit of 6 Mpa. The corresponding stress in the steel would be

 $\sigma_{st} = 14.286(6) = 85.72 \text{ MPa}$ 

which is below the allowable stress of 120 MPa.

The maximum safe axial load is thus found by substituting  $\sigma_{co} = 6$  MPa and  $\sigma_{st} = 85.72$  MPa (rather than  $\sigma_{st} = 120$  MPa) into the equilibrium equation :

$$P = \sigma_{st}A_{st} + \sigma_{co}A_{co}$$
  
= (85.72 × 10<sup>6</sup>) (3.6 × 10<sup>-3</sup>) + (6 × 10<sup>6</sup>) (86.4 × 10<sup>-3</sup>)  
= 827 × 10<sup>3</sup> N = 827 kN Answer

#### **ADDITIONAL PROBLEMS**

#### Problem 1

A reinforced concrete column 200 mm in diameter is designed to carry an axial compressive load of 300 kN. Determine the required area of the reinforcing steel if the allowable stresses are 6 MPa and 120 MPa for the concrete and steel, respectively. Use Eco = 14 GPa and Est = 200 GPa.

Solution 234

$$\delta_{co} = \delta_{st} = \delta$$

$$\left(\frac{PL}{AE}\right)_{co} = \left(\frac{PL}{AE}\right)_{st}$$

$$\left(\frac{\sigma L}{E}\right)_{co} = \left(\frac{\sigma L}{E}\right)_{st}$$

$$\frac{\sigma_{co}L}{14000} = \frac{\sigma_{st}L}{200\,000}$$

$$100\sigma_{co} = 7\sigma_{st}$$

When  $\sigma_{st} = 120$  MPa  $100\sigma_{co} = 7(120)$ 

 $\sigma_{co} = 8.4 \text{ MPa} > 6 \text{ MPa} \text{ (not ok!)}$ 

 $\begin{array}{l} \mbox{When } \sigma_{co} = 6 \mbox{ MPa} \\ 100(6) = 7 \sigma_{st} \\ \sigma_{st} = 85.71 \mbox{ MPa} < 120 \mbox{ MPa} \mbox{ (ok!)} \end{array}$ 

Use  $\sigma_{c0} = 6$  MPa and  $\sigma_{st} = 85.71$  MPa



$$\begin{split} \Sigma F_V &= 0\\ P_{st} + P_{co} &= 300\\ \sigma_{st} A_{st} + \sigma_{co} A_{co} &= 300\\ 85.71 Ast + 6 \left[\frac{1}{4}\pi (200)^2 - A_{st}\right] &= 300(1000)\\ 79.71 A_{st} + 60\ 000\pi &= 300\ 000\\ A_{st} &= 1398.9\ \text{mm}^2 \rightarrow answer \end{split}$$

#### **MECHANICS OF MATERIALS**

#### THANKS

## **Normal Strain and Stress**

Normal Strain and Stress, Stress strain diagram, Hooke's Law

# Strain

- When a body is subjected to load, it will deform and can be detected through the changes in length and the changes of angles between them.
- The deformation is measured through experiment and it is called as strain.
- The important of strain: it will be related to stress in the later chapter



## **Normal Strain**

Normal strain is detected by the changes in length.

$$\mathcal{E} = \frac{l'-l}{l}$$

$$= \frac{\Delta l}{l}$$

$$\mathcal{E} \text{ (epsilon)}$$

$$l': \text{ length after deformed}$$

$$\exists \text{ l: original length.}$$

Note  $\epsilon$ :

- dimensionless
- very small (normally is  $\mu m$  (=10<sup>-6</sup> m))
- 480(10)<sup>-6</sup> m/m = 480 μm/m = 480 "micros" = 0.0480 %



When load P is applied, the <u>RIGID</u> lever arm rotates by 0.05°. Calculate the normal strain of wire BD

Foundation: △L/L Knowledge required: geometrical equation Rigid: no deformation on the lever

## **Geometry: The mathematics**

Sine and Cosine Rule

$$L1 = \sqrt{L2^2 + L3^2 - 2(L2)(L3)\cos(\gamma)}$$



When force P is applied to the rigid lever arm

 $L_{BD}$  after deformed is DB' Cosine rule can be applied here

$$L_{BD'} = \sqrt{L_{AB'}^{2} + L_{AD}^{2} - 2(L_{AB'})(L_{AD})\cos(\phi + 0.05)}$$



Strain:

$$\varepsilon_{BD} = \frac{L_{DB'} - L_{DB}}{L_{DB}}$$

When force P is applied to the rigid lever arm

L<sub>BD</sub> after deformed is DB' Cosine rule can be applied here

 $L_{BD'} = \sqrt{L_{AB'}^{2} + L_{AD}^{2} - 2(L_{AB'})(L_{AD})\cos(\phi + 0.05)}$ 

$$L_{BD'} = 300.3491 \, mm$$

Strain:

$$\varepsilon_{BD} = \frac{L_{DB'} - L_{DB}}{L_{DB}}$$

$$\varepsilon_{BD} = 0.00116 \, mm / mm$$



The force applied to the handle of the rigid lever the arm to rotate clockwise through an angle of 3° about pin A. Determine the average normal strain developed in the wire. Originally, the wire is unstretched.

D 600 mm A 45° B

Discuss the approach?

Prob. 2-21

## Solution







ε= 0.0258 m/m

# **Simple Tensile Test**

- Strength of a material can only be determined by *experiment*
- The test used by engineers is the tension or compression test
- This test is used primarily to determine the relationship between the average normal stress and average normal strain in common engineering materials, such as metals, ceramics, polymers and composites





### **Conventional Stress–Strain Diagram**

 Nominal or engineering stress is obtained by dividing the applied load P by the specimen's original cross-sectional area.

$$\sigma = \frac{P}{A_0}$$

 Nominal or engineering strain is obtained by dividing the change in the specimen's gauge length by the specimen's original gauge length.

$$\varepsilon = \frac{\delta}{L_0}$$

#### **Conventional Stress–Strain Diagram**



Conventional and true stress-strain diagrams for ductile material (steel) (not to scale)

## **Conventional Stress–Strain Diagram**

#### **Elastic Behaviour**

- A straight line
- Stress is proportional to strain, i.e., linearly elastic
- Upper stress limit, or *proportional limit*;  $\sigma_{pl}$
- If load is removed upon reaching elastic limit, specimen will return to its original shape

#### Yielding

- Material deforms permanently; yielding; plastic deformation
- Yield stress,  $\sigma_Y$
- Once yield point reached, specimen continues to elongate (strain) without any increase in load
- Note figure not drawn to scale, otherwise induced strains is 10-40 times larger than in elastic limit
- Material is referred to as being *perfectly plastic*

## **Hooke's Law**

• *Hooke's Law* defines the *linear relationship* between stress and strain within the elastic region.

- ctrocc

$$\sigma = E \mathcal{E}$$
  $E = modulus of elasticity or Young's modulus$   
 $\varepsilon = strain$ 

• *E* can be used only if a material has *linear*elastic behaviour.

E can be derived from stress and strain graph. What is it?



- The stress-strain diagram for an aluminum alloy that is used for making aircraft parts is shown. When material is stressed to 600 MPa, find the permanent strain that remains in the specimen when load is released. Also, compute the modulus of resilience both before and after the load application.
- Approach to the problem:
  - Parallel to elastic line
  - Both slope is equal
  - Distance CD can be calculated based on the slope
  - Permanent strain: 0.023 distance CD



## Solution

- When the specimen is subjected to the load, the strain is approximately 0.023 mm/mm.
- The slope of line OA is the modulus of elasticity,
- From triangle *CBD*,

$$E = \frac{450}{0.006} = 75.0 \,\mathrm{GPa}$$

$$E = \frac{BD}{CD} = \frac{600(10^6)}{CD} = 75.0(10^9) \Longrightarrow CD = 0.008 \text{ mm/mm}$$

The bar DA is rigid and is originally held in the horizontal position when the weight W is supported from C. If the weight causes B to be displaced downward 0.625mm, determine the strain in wires DE and BC. Also if the wires are made of A-36 steel and have a crosssectional area of 1.25 mm<sup>2</sup>, determine the weight W.





Prob. 3-42

1) Calculate the displacement of D

$$\frac{\delta_D}{1.5} = \frac{\delta_B}{0.9}$$
$$\delta_D = 0.625(\frac{1.5}{0.9})$$
$$\delta_D = 1.0417 mm$$



2) Based on displacement on D, calculate the strain and normal stress

$$\varepsilon_D = \frac{\delta_D}{L_D} = \frac{1.0417}{900} = 1.157(10)^{-3} \, mm/mm$$
$$\sigma_D = E\varepsilon = 200(10)^3 1.157(10)^{-3} = 231.4 MPa$$

\* strain in mm/mm, stress and E in MPa, F in N and length in mm 3) Based on normal stress at wire DE, calculate the T of wire D

$$\sigma_D = 231.4MPa = \frac{T_{ED}}{A}$$
$$T_{ED} = \sigma_D A = 289.3N$$



4) Calculate W, based on FBD of bar DA

$$\sum M_{A} = 0$$
  
- T<sub>DE</sub>(1.5) + W(0.9) = 0  
W = 482.2N

5) Calculate normal stress of wire CB and strain of wire CB

$$\sigma_{BC} = \frac{T_{BC}}{A} = \frac{482.2}{1.25} = 385.7 MPa$$

Strain can not be calculated as normal stress goes beyond yield stress (Sy = 250 MPa), elastic property is no more applied. Therefore it requires the stress and strain curve to predict the strain